

# Time Response Analysis of Control System

## 4

The time response of a control system means as to how a system behaves in accordance with time when a specified input test signal is applied.

The time response of a control system is divided in two parts:

- (i) Transient response
- (ii) Steady state response

Transient part of time response reveals the nature of response and its speed, where as steady state part of time response reveals the accuracy of a control system.

### Standard Input Test Signals

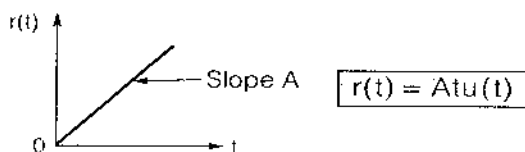
#### (a) Step function



where,

$$u(t) = \begin{cases} 1 & ; \quad t \geq 0 \\ 0 & ; \quad t < 0 \end{cases}$$

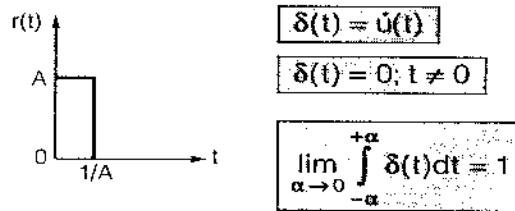
#### (b) Ramp function



#### (c) Parabolic function



### (d) Impulse Function



### Time response of a first order control system

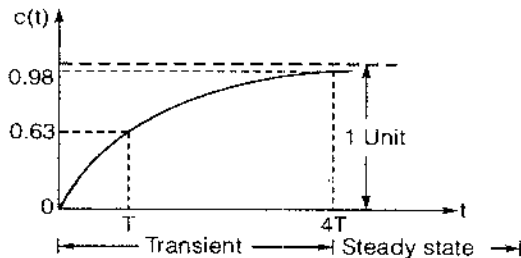


#### □ Transfer Function

$$\frac{C(s)}{R(s)} = \frac{1}{sT + 1}$$

### (i) Subjected to Unit Step Input Function

$$C(s) = \frac{1}{s(sT + 1)}$$



#### □ Time response expression

$$c(t) = \mathcal{L}^{-1}C(s) = 1 - e^{-t/T}$$

#### □ Error

$$e(t) = r(t) - c(t) = e^{-t/T}$$

#### □ Steady state error

$$e_{ss} = \lim_{t \rightarrow \infty} e^{-t/T} = 0$$

**Note:** .....  
 Lower the time constant, faster is the time response of a control system.  
 .....

### Subjected to Unit Ramp Input Function

#### □ Time response expression

$$c(t) = (t - T + Te^{-t/T})$$

#### □ Error

$$e(t) = r(t) - c(t) = (T - Te^{-t/T})$$

#### □ Steady State Error

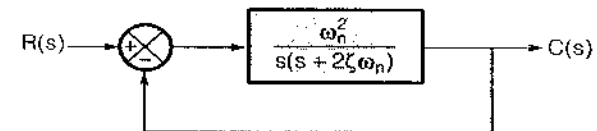
$$e_{ss} = \lim_{t \rightarrow \infty} (T - Te^{-t/T}) = T$$

### Subjected to Unit Impulse Input Function

#### □ Time response expression

$$c(t) = \frac{1}{T} e^{-t/T}$$

### Time Response of Second Order Control System



#### □ Transfer function

$$\frac{C(s)}{R(s)} = \frac{\omega_n^2}{s^2 + 2\zeta\omega_n s + \omega_n^2}$$

where,  $\omega_n$  = Natural frequency of oscillations

$\zeta$  = Damping ratio

$\zeta\omega_n$  = Damping factor

#### □ Damped frequency

$$\omega_d = \omega_n \sqrt{1 - \zeta^2}$$

where,  $\omega_d$  = Damped frequency of oscillations

## Subjected to Unit Step Input Function

### Time response expression

Case-1:  $\zeta < 1$  i.e. underdamped oscillations

$$c(t) = 1 - \frac{e^{-\zeta\omega_n t}}{\sqrt{1-\zeta^2}} \sin(\omega_d t + \phi)$$

$$\phi = \tan^{-1} \left( \frac{\sqrt{1-\zeta^2}}{\zeta} \right)$$

### Note:

- Response settles within 2% of the desired value (1 unit) after damping out the oscillations in a time  $4T$  (or  $4/\zeta\omega_n$ ).

Case-2:  $\zeta = 0$  i.e. sustained (undamped) Oscillations

$$c(t) = (1 - \cos \omega_n t)$$

Case-3:  $\zeta = 1$  i.e. critically Damped Oscillations

$$c(t) = [1 - e^{-\omega_n t}(1 + \omega_n t)]$$

Case-4:  $\zeta > 1$  i.e. overdamped Oscillations

$$c(t) = 1 - \frac{e^{-(\zeta - \sqrt{\zeta^2 - 1})\omega_n t}}{(2\sqrt{\zeta^2 - 1})(\zeta - \sqrt{\zeta^2 - 1})}$$

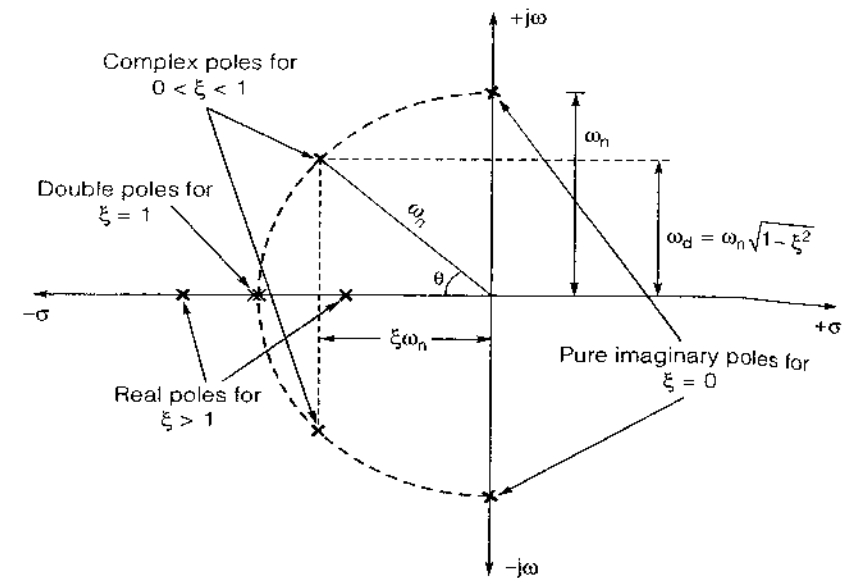
### Time constant of the response

$$T = \frac{1}{(\zeta - \sqrt{\zeta^2 - 1})\omega_n}$$

## Effect of $\xi$ on Time Response

SNo.	Value of $\xi$	Type of closed poles	Nature of response	System classification
1.	$\xi = 0$	Purely imaginary	Oscillations with constant frequency and amplitude	Undamped
2.	$0 < \xi < 1$	Complex conjugate	Damped oscillation	Underdamped
3.	$\xi = 1$	Real, equal and negative	Critical and pure exponential	Critically damped
4.	$1 < \xi < \infty$	Real, unequal and negative	Purely exponential slow and sluggish	Overdamped

## Pole Locations for Different Cases, for a Second Order System



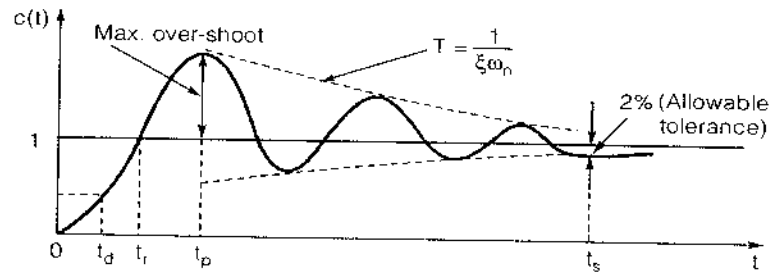
### Note:

- As  $\xi$  increases from 0 to 1 then the frequency of oscillation reduces.
- For all positive value of  $\xi$ , system are stable.
- Pole away from origin called insignificant pole, and does not affect the stability of system.
- $\theta = \cos^{-1} \xi$ .

## Some Practical Example of Second Order System

Series RLC Circuit	Parallel RLC Circuit	Translatory System	Rotational System
Characteristic equation $s^2 + \frac{R}{L}s + \frac{1}{LC} = 0$ $\omega_n = \frac{1}{\sqrt{LC}}$ $\xi = \frac{R}{2\sqrt{L/C}}$	Characteristic equation $s^2 + \frac{1}{RC}s + \frac{1}{LC} = 0$ $\omega_n = \frac{1}{\sqrt{LC}}$ $\xi = \frac{1}{2R\sqrt{L/C}}$	Characteristic equation $s^2 + \frac{f}{M}s + \frac{K}{M} = 0$ $\omega_n = \sqrt{\frac{K}{M}}$ $\xi = \frac{f}{2\sqrt{KM}}$	Characteristic equation $s^2 + \frac{f}{J}s + \frac{K}{J} = 0$ $\omega_n = \sqrt{\frac{K}{J}}$ $\xi = \frac{f}{2\sqrt{KJ}}$

## Transient Response Specifications of Second Order Control System



### 1. Delay Time ( $t_d$ )

It is the time require for the response to rise from 0 to 50% of the final value.

$$t_d = \frac{1 + 0.7\xi}{\omega_n}$$

### 2. Rise Time ( $t_r$ )

It is the time required for the response to rise from

0 to 100% (underdamped) → for underdamped system

5 to 95% (critical damped) → for critically damped system

10 to 90% (over damped) → for overdamped system

of the final value

$$t_r = \frac{\pi - \tan^{-1}\left(\frac{\sqrt{1-\xi^2}}{\xi}\right)}{\omega_d} = \frac{\pi - \cos^{-1}\xi}{\omega_d} \text{ radian}$$

### 3. Peak Time ( $t_p$ )

It is the time required for the response to rise from zero to peaks of the time response.

$$t_p = \frac{n\pi}{\omega_d}$$

where,

$n = 1, 2, 3, \dots$

for

$n = 1$ , first overshoot

$n = 2$ , first undershoot

### 4. Settling Time ( $t_s$ )

It is the time require for response to rise and reach to the tolerance band

For 2% tolerance band,

$$t_s = 4\tau = \frac{4}{\xi\omega_n}$$

For 5% tolerance band,

$$t_s = 3\tau = \frac{3}{\xi\omega_n}$$

### 5. Peak Over Shoot ( $M_p$ )

It gives the normalised difference between the steady state output to first peak of the time response.

$$\%M_p = e^{-\left(\frac{\pi\xi}{\sqrt{1-\xi^2}}\right)} \times 100\%$$

**Note:**

- The time period of the oscillation before reaching the steady state.

$$T_{\text{oscillation}} = \frac{2\pi}{\omega_d}$$

- Number of oscillation before reaching steady state is

$$N = \frac{T_{\text{settling}}}{T_{\text{oscillation}}}$$

## Types and Order of System

- Every Transfer function representing the control system has certain type and order.
- The steady state analysis depends on type of the system.
- The type of the system is obtain from open loop transfer function i.e.  $G(s) H(s)$ .
- The number of open loop poles occurring at origin determines the type of the system.

$$\text{Let } G(s) H(s) = \frac{K(1+sT_a)}{s^P(1+sT_1)}$$

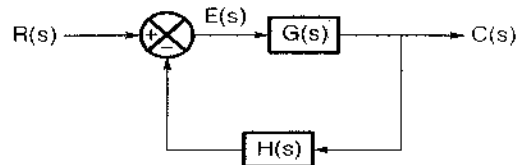
For  $P = 0$  for Type-0 system  
 $P = 1$  for Type-1 system  
 $P = n$  for Type-n system

- Transient state analysis depends on order of the system.
- The order of system is obtain from closed loop transfer function i.e.

$$\frac{G(s)}{1 + G(s)H(s)}$$

- The highest power of characteristic equation, i.e.  $1 + G(s) H(s) = 0$  determines the order of control system.

## Steady State Error



$$E(s) = \frac{R(s)}{1 + G(s)H(s)}$$

Steady state error,  $e_{ss} = \lim_{t \rightarrow \infty} e(t)$  (in time-domain)

$$e_{ss} = \lim_{s \rightarrow 0} sE(s) \quad (\text{in s-domain})$$

$$e_{ss} = \lim_{s \rightarrow 0} s \frac{R(s)}{1 + G(s)H(s)}$$

$$e_{ss} = \frac{\lim_{s \rightarrow 0} sR(s)}{1 + \lim_{s \rightarrow 0} G(s)H(s)}$$

**Note:**

Steady state error depends on two factor:

- (i) Type of input applied i.e.  $R(s)$ .
- (ii) Type of system i.e.  $G(s) H(s)$ .

### 1. Steady State Error for Different Types of Input

#### (i) Step input

$$R(s) = \frac{A}{s}$$

$$e_{ss} = \frac{A}{1 + K_p}$$

$$K_p = \lim_{s \rightarrow 0} G(s)H(s)$$

where,  $K_p$  = Static position error constant

#### (ii) Ramp input

$$R(s) = \frac{A}{s^2}$$

$$e_{ss} = \frac{A}{K_v}$$

$$K_v = \lim_{s \rightarrow 0} sG(s)H(s)$$

where,  $K_v$  = Static velocity error constant

**(iii) Parabolic input**

$$R(s) = \frac{A}{s^3}$$

$$e_{ss} = \frac{A}{K_a}$$

$$K_a = \lim_{s \rightarrow 0} s^2 G(s) H(s)$$

where,  $K_a$  = Static acceleration error constant

**2. Steady State Error for Different Types of System**

Type	Step input	Ramp input	Parabolic input
Type-0	$\frac{A}{1+K}$	$\infty$	$\infty$
Type-1	0	$\frac{A}{K}$	$\infty$
Type-2	0	0	$\frac{A}{K}$

Where, K is the system gain.

**Observation**

- (i)  $e_{ss} \propto \frac{1}{K}$ , so, as system gain increases, steady state error decreases.
- (ii) For Linear Time Invariant (LTI) system, the maximum type number is two. Beyond type-2, the system exhibits non-linear characteristics.

**Note:**

- Steady state error is valid only for closed loop stable system'
- Steady state error are calculated to the closed loop system by using open loop transfer function.